NASA Technical Memorandum 80075

PERFORMANCE ESTIMATION FOR A HIGHLY LOADED EIGHT-BLADE PROPELLER COMBINED WITH AN ADVANCED TECHNOLOGY TURBOSHAFT ENGINE

SHELBY J. MORRIS, JR.

APRIL 1979



PERFORMANCE ESTIMATION FOR A HIGHLY LOADED EIGHT-BLADE PROPELLER COMBINED WITH AN ADVANCED TECHNOLOGY TURBOSHAFT ENGINE

Shelby J. Morris, Jr. Langley Research Center

SUMMARY

Performance estimation, weights, and scaling laws for an eight-blade highly loaded propeller combined with an advanced turboshaft engine are presented. The data are useful for planned aircraft mission studies using the turboprop propulsion system. Comparisons are made between the performance of the 1990+ technology turboprop propulsion system and the performance of both a current technology turbofan and an 1990+ technology turbofan.

INTRODUCTION

Recent predicted improvements in the propulsive efficiency of highly loaded propellers at cruise Mach numbers of 0.8 have led to increased interest (ref. 1) in the use of these devices to propel advanced aircraft. Early studies indicate that, compared to a high bypass turbofan, turbine-propeller systems offer a potential reduction of 15 to 30 percent in fuel consumption, a reduction in direct operating cost of approximately 10 percent, and a reduction in aircraft gross weight of approximately 20 percent for long endurance missions. The recent increased emphasis on reduction in fuel consumption created by decreasing supplies and increasing cost of fuel make this concept very timely. While near field noise, passenger comfort, and maintenance remain as potential problems, the possible benefits of this concept have lead to system studies of possible future propeller powered aircraft. These system studies require that the weight, scaling, and performance of the concept be known.

The purpose of this report is document the predicted performance, scaling, and weight of an eight-blade highly-loaded propeller (propfan) combined with an advanced turboshaft engine (the Pratt and Whitney STS 487 of reference 2), and to compare these results directly with an advanced high-bypass ratio turbofan. The approach used herein combines predicted propeller data with the engine characteristics to yield the uninstalled performance of the propeller-engine combination in terms identical to those by which turbofan performance characteristics are generally presented.

Few large commercial propeller-driven aircraft have been designed in the last two decades; thus, computer programs equivalent to those for jet aircraft often do not exist for sizing and predicting the performance of turboprop aircraft. The present paper provides a means of converting propeller and engine data to a common basis with turbofan engines. The converted data may be used directly in available programs, thus eliminating the need for developing new computer software.

SYMBOLS

```
propeller power coefficient. (P/D^2)/\rho(ND)^3
C_{\mathbf{p}}
          propeller thrust coefficient, T/(\rho(ND)^2 D^2)
\mathsf{C}^{\mathsf{T}}
          propeller diameter, m(ft)
D
          gravitational constant, 1.0(32.2 \text{ ft/s}^2)
g
           gearbox gear ratio
GR
          fuel lower heat value; 42.717 \times 10^6 J/kg (18,400 Btu/lbm)
Н۷
           propeller advance ratio, (V/ND)
J
JC
           conversion factor: 1.0(778 ft lbf/Btu)
Κ<sub>1</sub>
           constant in equation (10); 0.6325 (0.0935)
           constant in equation (11); 1.0203 \times 10^{-2} (5.115 x 10^{-3})
K<sub>2</sub>
           constant in equation (12); 6.3483 \times 10^{-2} (0.1044)
Ka
           mass, kg (1bm)
М
           propeller revolutions per unit time, revolutions/s
N
Р
           engine shaft power, kw (hp)
T
           net uninstalled thrust, N (1bf)
           thrust specific fuel consumption, (kg/hr)/N ((lbm/hr)/lbf)
TSFC
٧
           velocity, m/s (ft/s)
           weight, N (lbf)
W
```

W_f fuel flow, kg/s (lbm/s)

 ρ density, kg/m³ (slugs/ft³)

η efficiency

Subscripts:

G/B gearbox

jet jet exhaust

net net uninstalled

NOM nominal

o free stream

ov overall

p propulsive

prop propeller

t thrust

tip propeller tip

RESULTS AND DISCUSSION

The Pratt and Whitney STS 487 turboshaft data (ref. 2) was matched with eight-blade propfan data (ref. 3) to calculate the uninstalled net thrust, fuel flow, thrust specific fuel consumption, and overall efficiency of the combined propulsion system. These parameters, and other associated engine and propeller parameters, are obtained as functions of Mach number, altitude, and power setting for both standard day and nonstandard day atmospheres. The propeller is first sized at a selected operating condition. The sizing data required are the design values of Mach number, altitude, shaft horsepower, residual jet thrust, propeller tip speed, power coefficient, and delta temperature above a standard day.

First, the free stream velocity and density are obtained for the known Mach number, altitude and ambient temperature, then the propeller diameter can be obtained as

$$D = \left[\frac{P}{C_{p} \rho(ND)^{3}}\right]^{1/2} \tag{1}$$

where

$$ND = V_{tip}/3.1416$$
 (2)

The advance ratio (J) is

$$J = V/ND \tag{3}$$

The thrust coefficient C_{T} can be obtained from the tables of reference 3 as a function of Mach number, advance ratio, and power coefficient. Finally, the propeller thrust can be computed from the thrust coefficient and the results of equations (1) and (2) as

$$T_{prop} = C_T \rho(ND)^2 D^2 \tag{4}$$

The net thrust can be calculated by adding the residual jet thrust, which is obtained from the engine specifications to the net propeller thrust

$$T_{\text{net}} = T_{\text{prop}} + T_{\text{jet}}$$
 (5)

The fuel flow is known from the engine data of reference 2, and when combined with the results of equation (5), the equivalent thrust specific fuel consumption is calculated as

$$TSFC = W_f/T_{net}$$
 (6)

Finally, the efficiencies are obtained from their definitions (ref. 4) as

$$\eta_{\text{oV}} = \frac{T_{\text{net}} V_{\text{o}}}{W_{\text{f}} \text{ JC HV} + V_{\text{o}}^2/2g}$$
 (7)

$$n_{p} = \frac{C_{T}J}{C_{p}} \tag{8}$$

For the off-design cases, the propeller diameter is fixed from the design computation and the power coefficient is computed from the known engine shaft power, the propeller diameter, the specified propeller rotational tip speed, and equation (2),

$$C_{p} = \frac{P/D^{2}}{\rho(ND)^{3}} \tag{9}$$

Then, the advance ratio $\, \, J \,$ is computed from equation (3) using the velocity corresponding to the specified Mach number, altitude, and temperature. Using these values of $\, \, C_p \,$ and $\, \, J \,$, and the tables of reference 3, the off-design thrust coefficient $\, \, C_T \,$ can be computed. Equations (4) to (8) are then used to compute the remaining off-design parameters.

The propeller, gearbox, and turboshaft engine weights are estimated using the methods of reference 3 for the propeller and gearbox and reference 2 for the turboshaft engine. Curve fits of the methods described in these references result in the following approximate expressions:

$$WT_{prop} = K_1 D^{2.48808} V_{tip}^{0.30} (P/D^2)^{0.2925}$$
 (10)

$$WT_{GB} = K_2 \left[\frac{GR}{8}\right]^{1/2} \left[\frac{P}{D^2}\right] D^3$$
 (11)

$$WT_{FNG} = K_3 (P/D^2) D^2$$
 (12)

The total uninstalled weight is the sum of equations (10), (11), and (12). It is recommended in reference 3 that this uninstalled weight be multiplied by 1.3 to account for installation. The nominal engine and propeller sizes can be scaled using the equations of references (2) and (3) (Appendix A). It is recommended that the engine scale factor be limited to the range between 0.7 and 1.45. In equation (10, (11), and (12), the power loading (P/D^2) has been expressed separately because, for a given design Mach number and altitude, the power loading (or power coefficient) is held constant as the engine size is scaled to match the required mission thrust. Furthermore, at the design Mach

number and altitude, and a constant specified tip speed, a constant power loading (P/D^2) uniquely fixes the advance ratio, the power coefficient, the thrust coefficient, the propulsive efficiency, the overall efficiency, and the thrust specific fuel consumption as the engine is scaled.

Sample Calculation

This procedure will now be illustrated by some numerical results which assumes that the eight-bladed propeller of reference 3 is matched to the turboshaft engine of reference 2. The tip speed is to be held constant at 244 m/s (800 ft/s). The propeller is to be sized for a Mach number of 0.8 at 11 km (36,089 ft). The propeller power coefficient at the design point is selected to be 1.692 with P/D^2 of 281 kw/m² (35 hp/ft²). The propeller propulsive efficiency is 0.817 at the design conditions. The resulting design-size information is summarized in Table 1. Typical sea-level-static performance is also shown in Table I. The resulting propeller diameter for the baseline size of 15,238 kw (20,438 hp) at sea-level-static maximum-power condition was 4.519 m (14.825 ft). Note that the data from Table I indicates that the propeller was not able to absorb all the power available at sea level static conditions because of propeller stall. This phenomena is also shown in figure 1 which presents the thrust specific fuel consumption versus the uninstalled net thrust for the nominally sized propfan propulsion system. The total weight of the uninstalled engine, gearbox, and propeller using the nominal 15,238 kw (20,438 hp) engine size is estimated at 2653 kg (5851 1bm). The propeller tip speed of 244 m/s (800 ft/s) results is a propeller which has 1,030 revolutions per minute and a gearbox gear ratio of approximately 8.25.

Figure 1 summarizes the performance of the nominally sized propfan propulsion system. This figure presents the uninstalled thrust specific fuel consumption versus uninstalled net thrust for several different Mach numbers and altitude combinations.

A comparison of the turboprop (propfan) performance and the performance of the Pratt and Whitney JT9D-25 turbofan (ref. 5) is shown in figure 2. The turboprop has been scaled to match the maximum thrust of the JT9D-25 at Mach number 0.8 and 9.144 km (30,000 ft.). The JT9D-25 engine represents 1965 to 1970 technology while the turboprop gas generator represents 1990+ technology. Data for the 1990+ technology Pratt and Whitney STF-477 turbofan (ref. 6) has also been scaled to the same conditions and is shown in figure 2 to provide a more valid comparison of systems at the identical level of technology. It can be seen from figure 2 that the turboprop propulsion system reduces the cruise thrust specific fuel consumption at a Mach number of 0.8, 30,000 feet by about 27 percent when compared with the 1965 to 70 technology JT9D-25, and by about 22 percent compared with the 1990+ technology STF-477 turbofan.

A comparison of selected engine performance parameters for each of these scaled propulsion concepts are shown in Table II. Estimated weight scaling laws for the turboprop (propfan) engine concept are shown in figure 3. Note that the

turboprop weights increase more rapidly with size increase than does a typical turbofan which are also shown on this figure. This difference would impose an increased penalty on the turboprop as it was scaled upward and would tend to drive aircraft configurations towards a large number of small engine rather than a small number of large engines. The same conclusion may be reached by considering how the weights for the propeller and gearbox scale using equations (10) and (11), respectively. Of course, cost, complexity and availability would also be considerations in any choice of propulsion systems.

A summary of the performance data for the eight-blade turboprop (propfan) combined with an advanced turboshaft engine is presented as Appendix B.

CONCLUDING REMARKS

Performance estimations, weights, and scaling laws for an eight-blade high-loaded propeller combined with an advanced turboshaft engine have been presented. The data is useful for planned aircraft mission studies using this propulsion system. Comparisons are made between the performance and weight of the 1990+technology turboprop propulsion system and the performance and weight of a current technology turbofan and a 1990+technology turbofan.

At Mach numbers of 0.8 and 9,144 meter altitude, the 1990+ technology turboprop produced thrust specific fuel consumption values about 27 percent less than the current technology turbofan and about 22 percent less than the 1990+ advanced technology turbofan. These benefits must be measured against the increased system weight, potential increased maintenance problems, and noise and passenger acceptance considerations before any conclusions can be made.

This report proports to furnish the necessary propulsion data to enable evaluation of air transportation systems using these turboprop propulsion system. The data is presented in a format compatible with existing mission programs.

REFERENCES

- 1. Dugan, J. F.; Gatzen, B. S.; and Adamson, W. M: Propfan Propulsion Its Status and Potential SAE Paper Number 780995, November 27-30, 1978.
- 2. Kropp, W. J.: Preliminary Performance and Installation Data for the STS-487 Turboshaft Engine. Pratt and Whitney Aircraft Report CDS-11 East Hartford, CN (Some of the data in the report is Proprietary.).
- 3. Baum, J. A. et al: Propfan Data Support Study Technical Report. NASA CR-152141, Hamilton Standard Division of United Technologies Corporation, Windsor Locks, CN, February 28, 1978.
- 4. Shepard, D. G.: Aerospace Propulsion, American Elsevier Publishing Company, Inc., New York, NY 1972.
- 5. USAF Propulsion Characteristics Summary, Volume 1 Airbreathing. Aeronautical Systems Division, Air Force Systems, Command Wright Patterson Air Force Base, OH June 1972. (Some items in this report are classified Confidential.).
- 6. Perkins, G. M.: Preliminary Performance and Installation Data for the STF 477 Turboshaft Engine. Pratt and Whitney Aircraft Report CDS-6, East Hartford, CN (Some of the data in this report is Proprietary.).

Appendix A

SCALING LAWS FOR TURBOPROP ENGINE

Engine:

Nominal Engine Length = 2.240 m (7.350 ft) Nominal Engine Maximum Diameter = 0.914 m (3 ft) Nominal Engine RPM = 8500Nominal Engine Takeoff Power = P_{NOM} = 15.23 Mw (20424 hp)

Configuration:

Two spool gas generator with free turbine and third coaxial shaft technology level 1990+ (in service)

Scaling:

$$D_{eng} = \left(\frac{P_{Takeoff}}{P_{NOM}}\right)^{0.5}$$

$$D_{eng}_{NOM}$$

$$Length_{eng} = \left(\frac{P_{Takeoff}}{P_{NOM}}\right)^{0.43}$$

$$Length_{eng}_{NOM}$$

Propeller

$$D = D_{NOM} \left[\frac{P_{Takeoff}}{P_{NOM}} \right]^{1/2}$$

Free tubine RPM

$$RPM = RPM_{NOM} \left[\frac{P_{NOM}}{P_{Takeoff}} \right]^{0.5}$$

Appendix B

TYPICAL PROPFAN PERFORMANCE OUTPUT

Eight-Blade Propeller

SIZE POINT

Mach Number	=	0.80
Altitude	=	11 km (36089 ft)
Delta Temperature	=	10 ^o F `
J	=	3.081
Č _p	=	1.692
c_{T}	=	0.4485
η _p	=	0.8167

DEFINITIONS FOR COMPUTER OUTPUT

ΧM Mach number altitude (feet) ALT delta temperature above standard day (OF) DELT SHP engine shaft horsepower (hp) Τ propeller net thrust (lbf) T + FRES propeller + residual jet net thrust (lbf) propulsive efficiency (C_TJ/C_p) **ETAP** $C_{\mathbf{p}}$ propeller power coefficient propeller thrust coefficient C_T propeller advance ratio J WFT engine fuel flow (1bm/hr)

TYPICAL PROPFAN PERFORMANCE OUTPUT

ERROR MESSAGE "ERROR CP OUT OF RANGE OF TABLE" indicates that the engine is producing more power at the considered power setting than the propeller can absorb. NOTE:

<u> </u>	58.08	4.00	13.18 01.46 25.72 94.92	3.0 0.8	80.25 68.06 119.77 88.04 21.41 13.26 31.81 28.33 15.43	44.78 06.16 48.09
3	625	10 to 01	37 22 22 22	9	らちょよきさらまま ろうらりらりをすることできると	40 m
~	0.000	000	000000000000000000000000000000000000000	000		.877 .877
13	0000000	000	.5539 .5391 .5051	398 339	. 7208 . 6600 . 6600 . 5800 . 7323 . 4712 . 2412	.6556
d		.148 .001 .856	.7213 .5976 .4815	289 219	1.3112 1.1526 1.0081 .8602 .7243 .6012 .4837 .2915	1.3315
ETAP	0000.0	000	00000	0000	.2610 .2870 .3162 .3510 .3510 .4271 .5036	.4317
T+FRES		1665. 7752. 9961.	19750.1 19067.6 17752.4 15941.9	3888. 1792.	26157.0 24849.4 23589.7 22019.1 20458.8 18681.2 16460.3 14030.9	23739.4
-	0.0	0. 6335. 8780.	18777.4 18275.0 17122.9 15452.0	3502. 1498.	24432.3 23402.6 22371.7 21028.5 19659.0 18045.0 15973.6 11352.8	22221.7
SHP	TABL 0438.	F TABLE 18001.4 15694.7 13421.4	1303.5 9364.9 7545.0 5907.5	540.2	20547.6 18062.7 15798.3 13480.6 11350.5 17580.1 5939.5 4568.8	20866.2 18268.2
DELT	0 •		0000	• •	000000000	000
ALT	OUT OF		0000		0000000000	000
	م 2	d U				
Σ×	□ °	ERRUR, 0.00 0.00	0000	00	000000000000000000000000000000000000000	.20

FF	3748.02 3230.59 2752.28 2316.58 1945.43	6442.07 5682.76 5008.37 4366.21 3264.89 2342.95 1969.94	WFT 5465.86 5009.87 4404.92 3876.79 3365.56
7	. 877 . 877 . 877 . 877 . 877 . 778		0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
CT	.4956 .4387 .3768 .3117 .2514		0.0000 0.0000 0.0000 0.5390 8559
۵ ن	.7341 .6084 .4915 .3848 .2976	1.3632 1.1918 1.0412 .8888 .7500 .5214 .5028 .3946	CP 1.3219 1.2160 1.0566 .9169 .7739
ETAP	.5919 .6323 .6721 .7102 .7627	.5796 .6189 .6555 .7309 .7596 .7596 .8123 .8216	ETAP 6.0000 0.0000 0.0000 0.0000 0.0000
T+FRES	17431.6 15352.3 13116.2 10804.6 8676.1	21683.4 20059.1 18425.2 16533.7 14598.7 112495.8 10398.8 8377.9	T+FRES 1741.4 1573.7 1324.5 16867.5 17161.6
-	16799.0 14872.1 12771.5 10565.4 8522.6 6659.4	20363.9 19010.8 17589.8 15898.0 14127.5 12164.0 10192.5 6483.5 4996.7	T 0.0 0.0 15744.8 16238.1
SHP	11504.6 9534.2 7702.6 6030.6 4663.5 3539.4	21362.6 18676.9 16316.0 13927.4 11752.4 9737.1 7879.9 6183.9 4798.6	SHP 17849.7 17849.7 F TABLE 16420.6 F TABLE 14267.2 12381.3 10450.2 8741.0
OELT	000000		DELT RANGE D 0.0 RANGE D 0.0 0.0
ALT	000000	000000000	ALT F DUT DF 5000. F DUT DF 5000. 5000.
Σ ×	000000	0000000000	E X R O C O O O O O O O O O O O O O O O O O

Σ ×	ALT	DELT	SHP	-	T+FRES	ETAP	a o	C1	- 7	- H
0	000	•	7106.	5198.	5800	000	526	520	Ö	7 0 8 7
00.0	5000	0.0	5595	3931.	4406.	000	414	477		0000
0	000	•	4297.	2236.	2601.	000	318	41.8		751.0
0	000	•	3250.	10453.2	10738.6	0.0000	.2407	.3579	0000	
									ı	•
	000	•	17969	1205.	2751.	235	.330	726	43	400.0
-	000	•	16497.	0597.	1981.	248	.221	705	43	024.6
Н	000	•	14332.	9683.	0829.	273	061	673	. 4	416.5
•10	5000	0.0	12434.	43.	19598.8	298	.920	638	43	886.3
~	000	•	10495.	7465.	8237.	331	777	597	43	374.8
-	000	•	8761.	6144.	6762.	367	648	552	43	917.6
,	000	•	7135.	4511.	4988.	405	528	496	43	494.5
-	000	•	5622.	2556.	2917.	555	416	429	43	103.6
	000	•	4314.	0483.	0760	484	319	358	4	756.4
-	000	•	3272.	8544	8740.	.5201	.2423	•2925	.431	1466.56
•20	5000.		310.	9295.	0681.	419	. 35	960	8	562.2
	000	•	6727.	8794.	. 4000	447	.238	643	8	7.090
2	000	•	4527.	7799.	8778.	488	075	609	8	451.3
2	000	•	2589.	6631.	7428.	526	.932	569	86	014.0
	ာ (0	10631.1	15178.5	15805.0	.5687	787	519	8	398.6
V	000	•	894.	3640.	4122.	610	658	467	86	941.8
	000	•	232.	1847.	2203.	652	535	405	86	511.3
	000	•	706.	882.	0134.	689	422	338	86	118.2
	000	•	395.	987.	154.	723	325	273	86	769.8
	000	•	338.	309.	418.	752	47	.2160	.862	1479.04
•30	5000	0.0	18909.7	17874.4	19114.0	.5648	1.4004	.6119	1.292	5688.80

T H A	5138.53	960.1	438.6	972.3	538.8	142.2	792.3	400.5			4688.87		4374.86		0.00%	466.I	041.4	626.2	252.4	908.6	590.6	1321.16	7111.7	394.9	3976.77
- 7	1.292	20.	.29	.29	.20	.29	• 2 •	• 59	7)	00000		000.0	Č	200	00.	00.	00.	00.	00.	00.	0.000	.423	42	2
CT	.5825	495	441	386	328	267	211	163	CT		000000		000000	Č		27.0	556	555	532	493	440	.3780	.7268	77	688
G D	1.2658	.951	804	672	547	433	334	255	a O		1.3181		1.2410	100	771.	700	831	969	569	453	349	.2634	1.3273	• 548	.128
ETAP	.5948	673	710	742	774	798	816	826	ETAP		0.000		0 0000 0	000		000	000	000.	0000	000000	.000	0	.2317	41	258
T+FRES	18052.6	5117.	3390.	1632.	821.	951.	248.	792.	T+FRES		1506.1		1409.6	252	4171	0704	4798.	4556.	3898.		1371.	9733.	19543.4	9074.	8331.
-	17014.0	4470.	2905.	1280.	585.	812.	176.	770.	-		0.0		o ·	•	- L 806	* TO 7 7	3921.	3840.	3316.	62.	1019.	6463	18196.7	7822.	7237.
SHP	17092.4	2842.	0859.	076.	393.	848.	520.	450.	I	F TABL	15256.	F TABL	14363.6 F TABLE	12995	1210.	• • • • • •	621.	034。	591.	250.	040	048.	15363.0	4420	3062.
DELT	0.0	•	•	•	•	•	•	•	_	GE	0.0	ш (0°0	0.0	•	•	•	•	•	•	•	•	0.0	•	•
ALT	5000.	00	000	000	00	000	000	00	_	00	10000.	TOO	10000 CP OUT OF	10000	0000		0000	000	0000	000	0000	0000	10000	000	000
E ×	.30		3		n		m		×	20	0.0	2 (FREDR.	0	0	, (•	.	.	0	0	0	•10		

F- U.S	23 3476.40 23 3049.06 23 2633.02 23 2257.77 23 1909.68 23 1325.70	46 4783.29 46 4453.73 46 4009.76 46 3068.49 46 2652.89 46 1923.06 46 1336.21	69 4899.55 69 4545.53 69 4048.19 69 3556.59 69 2688.58 69 2299.64
- ->	4444444		
CT	.6533 .6155 .52157 .5215 .3858	.6622 .6510 .6510 .5891 .5891 .4912 .2980	.51140 .56183 .56183 .51177 .4676
CP	.9732 .6975 .5722 .4548 .3569	1.3554 1.2735 1.1440 .9867 .8443 .7071 .5797 .5797 .2703	1.4025 1.1635 1.0099 1.0099 1.0099 7242 .5931
ETAP	.2841 .34119 .38574 .4653		. 5575 . 6506 . 6506 . 6885 . 7575
T+FRES	17258.0 16151.6 14930.8 13524.6 11862.1 9923.1	17795.1 17411.2 16651.5 15515.1 14249.8 12773.6 11149.2 7637.9	16520.7 15926.1 14856.9 13602.0 12198.3 10735.5
-	16357.4 15410.7 14337.4 13057.4 11445.9 9659.3	16578.5 16297.9 15704.0 14748.8 13634.2 12296.9 10789.7 9139.9	15421.8 14938.3 14052.7 12961.3 11706.8 10376.2
SHP	11263.6 9664.8 8072.6 6622.3 5264.4 4061.9	15687.1 14739.3 13240.6 11420.5 9772.1 8183.7 6709.9 5342.5 4127.7	16233.1 15188.9 13466.4 11689.1 9977.8 8382.5 6864.8
DELT	000000	000000000	00000000
ALT	100000 100000 100000 100000	100000 100000 100000 100000 100000	1000000 1000000 1000000 1000000
Σ×	000000000000000000000000000000000000000	000000000000000000000000000000000000000	

Σ×	ALT	DELT	SHP	· •	T+FRES	ETAP	G P	5	7	WFT
• 30	10000.	0•0	3224.2	4509.2	4550.5	.8207	.2786	.1801	1.269	1353.76
2 0		— =	SHP	-	T+FRES	ETAP	G P	13	7	FIN
000	\circ	0 .0 N	2880. TABL	0.0	1294.1	000000	1.3058	0.00000	0000	3983.22
0.00 ERROR,	15000. CP OUT OF	0 • 0 N G E	4	0.0	1219.1	000000	1.2380	000000	00000	3736.53
0 0	000	о ш	1421. TABL	0	1129.0	000000	1.1579	000000	000.0	3468.73
0	000	0.0	0116.	•	75.	.000	025	00	00	8.060
0	2000	•	705.	1688	2505	000000	.8825	.547	00.	713.3
0	2000	•	337.	840.	2515.	.000	743	554	00	355.2
0	200	•	039.	1550.	2095.	.000	612	541	00.	019.4
0	200	•	840.	0858.	91.	000	490	508	00	713.8
0	500	•	766.	785.	0120.	0000	381	458	0	435.6
o.	200	•	839.	470.	8724.	000•	287	.3970	000.0	1188.85
	500	•	2971.	5490.	6652.	229	.315	726	41	1,000
	500	•	2298.	5210.	6299.	37	.246	712	41	754.9
•10	15000.	0.0	11490.2	14872.0	15872.3	.2485	1.1649	0269.	.415	3482.12
	500	•	0156.	4584.	5134.	70	.029	699	41	098.1
	500	•	747.	3468.	4170.	295	886	631	41	721.2
	500	•	364.	2621.	3189.	329	746	591	41	359.4
	500	•	690	1585,	2033.	366	619	543	41	025.0
	500	•	8.70	0284.	0629.	405	493	482	41	719.4
	500	•	785.	786.	044.	445	383	411	41	438.8
	200	•	855.	218.	404	485	289	338	41	191.9

VFT	065.0 810.7	3118.76 2744.93 2376.80 2040.85	449.4	168.8 902.8 593.1	2781-64 2405-90 2067-07 1752-53 1467-74	WFT 6277.24 5554.93 4923.94
~	$\mathbf{c} \mathbf{c} \mathbf{c} \mathbf{c} \mathbf{c} \mathbf{c}$		60 60 60 60	20.4	11.246	00000
CT	662 652 639		324	617 601 579 526	4,4936 3796 31796 1968	0.0000 0.0000 0.0000
O D	343 273 186	1.0410 .8996 .7564 .6242	369	391 317 220 041	6395 6395 6395 6395 3033	CP 1.3528 1.1853
ETAP	409 425 447	5642 6672 6672 6674 6503	650	55 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	. 7394 . 7394 . 7713 . 7993 . 8139	ETAP 0.0000 0.0000 0.0000
T+FRES	5187. 4908. 4525.	13675.9 12655.9 11430.9 10096.8 8627.2	098. 614.	145. 720. 139.	11022 9724 8362 6954 5586 4284	T+FRES 1900.3 1605.5 1375.1
-	4131. 3929. 3639.	12942.8 12062.3 10965.1 9741.3	913. 494.	180. 836. 357.	10532.0 9356.0 8099.9 6774.6 5480.1	0 0 0 0 0 0
SHP	251. 560. 701.	10268.4 8873.9 7461.6 6157.5	843. 906.	724. 995. 037. 474.	7625.0 6308.2 5057.9 3948.4 2991.7	SHP F TABLE 20224.0 F TABLE 17719.6 F TABLE 15523.8
DELT	• • •	000000	• •		000000	DELT RANGE O 25.0 RANGE O RANGE O 25.0
ALT	5000 5000 5000	15000. 15000. 15000. 15000.	5000 5000	5000 5000 5000	15000. 15000. 15000. 15000.	ALT CP OUT OF CP OUT OF CP OUT OF
Σ ×	200	00000000000000000000000000000000000000	22			ж ж в в в в в в в в в в в в в в в в в в

	6 ← 0	. 0	~	י עי	7 O		· (h-	~	S	~	.	. ,~	~			~	6 0			
H H	4294.4 3739.2 3237.0	764.9 337.6	297.5	569.0	4304.1	747.8	239.9	770.7	343.1	351.4	616.0	975.2	334.6	773.6	261.6	793.1	2359.98	7 08 7	60100	5034.34
7	0000		675.	† t	t t	44	77	7 7	7 7	O.	0	808.	O	0	0	0	.898	76	34	1.347
CT	6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	515 470	.7280	070	2 2	588	541	484	417	659	632	598	553	504	449	389	.3240	604	566	. 5243
a	.8889 .7535 .6284	509 402	1.3595	047	892	756	629	511	404	.377	206	•	904	766	638	520	0	.405	.233	1.0787
ETAP	0000	0 0 0 0 0 0	.2404	707	315	349	386	454	463	429	470	. 5083	549	590	631	672	708	579	619	.6545
T+FRES	18813.2 18907.8 18360.0	,300. 5687.	25211.9	2046.	1267.	9815.	8143.	6133.	3867.	2792.	1648.	20332.7	8695.	6915.	.0665	2940.	0714.	0799.	19331.5	7753.
-	17664.9 17958.0 17586.3	5200°	23541.6	1575	0306.	9038.	7518.	5654.	3506.	1333.	0456.	19348.4	7912.	6300.	4523.	2604.	0476.	9544.	18327.9	6953.
SHP	13289.2 11265.1 9394.5	011.	20325.0	5601	3345	1314.	9415.	653.	043•	0595.7	036.1	5798.6	520.8	460.7	545.9	782.6	138.7	1004.4	18433.5	6126.0
DELT	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	າ ທ	25.0	, 10	5	ŝ	ŝ	ŝ	Š	0	5.0	2.0	ر ا	Š.	ر د د	5	5	5°0	25.0	5.0
ALT	0000	• •	• •		•0					°		•						•	•	-
E ×	0000	•	000	ı 🗂		~ (⊣	• 20	2		V	V (N (Ü	V	• 30		m

Σ×	ALT	DELT	SHP	-	T+FRES	ETAP	d O	C1	7	VFT
	5000	5.	2915.	2357.	3214.	565	.379	610	.27	923.5
n	5000	Š	1795.	1811.	2552.	592	.260	583	.27	575.7
	500	5.	245.	941.	525.	631	960	540	.27	139.R
n	5000	ŝ	8859.	0052.	0512.	670	.946	496	.27	754.7
$\boldsymbol{\omega}$	5000	5.	7451.	8935.	9279.	708	796	441	.27	385.3
• 30	15000.	25.0	6189.	769.	019.	742	661	383	.27	0.55.9
	5000	'n	5005.1	6554.3	2.1	.7741	.5347	.3237	1.279	1750.42
Σ×	T T	_	ī		У Ц О Ц +	-		-		1
· 60	;		0971	0174.	1457	20.7	338	50.0	, ,	1766
• 30	•	0.0	18354.8	829	43.	.6238	171	555	31	
C	•0	•	5970.	7376.	8178.	199	.019	512	.31	912.4
m	• •	•	3611.	5645.	6255.	869	.868	461	.31	282.3
	•0	•	1456.	3858.	4307.	735	731	408	.31	713.2
3	•	•	9473.	1903.	2212.	764	604	351	.31	195.6
m	•	•	602.	9876.	0074.	486	485	291	.31	712.9
B	•	•	.696	008.	111.	815	380	236	.31	285.7
	•0	•	630.	264.	297.	822	295	184	.31	922.5
m	•	•	512.	88	77	828	.2242	.1412	1.315	9.6
	•0	•	1541.	8226.	9306	686	.374	5	.75	438.0
4	•0	•	8892.	6808.	7640.	721	.205	495	.75	688.3
	•	•	6422.	342.	5965.	757	048	452	.75	991.1
	•	•	4027.	3796.	4240.	197	.895	407	.75	356.2
• 40	0	0.0	11802.5	11897.3	12184.6	.8173	.7532	.3510	1.754	3773.39
	•0	•	9737.	877.	0042.	822	621	291	• 75	238.8
	•0	•	7844.	061.	8123.	833	500	237	•75	753.3
	•	•	6184.	456.	436.	846	394	190	7.5	322.4

E ×	ALT	DELT	SHP	-	T+FRES	ETAP	O D	CT	- 7	FF
.40	0	0.0	4818.4	983	0	.8386	30	147	.75	956.3
	•0	•	683.	3715.7	05.	81	.2350		1.754	1641.01
Σ×	-	-	I	۰	u a u	~	٥	۰		u
4	00	9	9441.	6272	7341	4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	430	C - ろろし	. 72	760.4
.40	5000	0.0	17162.4	15096.9	15937.9	. 7008	, [ر ا ا	77	1000
	00	•	5033.	3893.	4539.	736	.113	475	. 72	514.0
	00	•	2903.	2516.	2987.	772	.955	428	.72	943.9
	00	•	0915.	1067.	1391.	867	808	378	.72	424.1
	00	•	082.	9300.	9510.	815	672	318	.72	951.0
	00	•	377.	663.	769.	827	546	262	.72	515.5
	00	•	838•	157.	176.	840	432	210	.72	122.0
	္မ	•	527.	798.	766.	844	335	164	.72	777.9
	0	•	468.	009	528.	827	2	.1233	1.723	1489.56
	00	•	0258.	4988.	5923.	736	. 500	513	.15	935.0
	8	•	7793.	3684.	4365.	765	.317	468	51.5	235.4
	9	•	5591.	2365.	2860.	789	154	423		612.P
• 50	5000	0.0	58.	10890.1	11213.1	.8118	.9893	.3728	2.154	4022 93
	00	•	1321.	430.	9617.	829	838	322	.15	6 767
	00	•	9433	982.	053.	842	869	273	5	7.600
	9	•	7671.	563.	543.	852	568	224	4	563.9
	00	•	085.	213.	123.	853	450	178	.15	163.2
	0	•	769.	026.	882.	840	353	137	.15	819.9
•	2000	•	.999	.900	842.	816	271	102	5	524.7
		RANGE D	F TABL)	} }	i i	
•	5000	0.0	753	0.0	-182.9	000000	.2039	000000	2.154	1269.57
Σ.	ALT	DELT		-	T+FRES	-	သ	 		F
. 40	10000.	•	4	14137.2	5128	.6567	1,4553	.5647	1.692	5011.72

¥	ALT	DELT	SHP	-	T+FRES	ETAP	a O	CT	7	FAX
4	0000	•	5531.	3510.	4358	680	146.	520	40	7 603
040	10000	0.0	544.	410	061.	716	170	407	. 0	025.5
4	0000	•	1797.	1323.	1818.	751	.019	452	6.0	554.7
4	0000	•	0016.	0100.	0440	788	.865	403	69	7.060
4	0000	•	8402.	8657.	8892.	808	726	345	649	672.9
4	0000	•	869.	210.	348.	821	593	288	\$	283.3
4	0000	•	458	815.	873.	833	471	232	69	927.3
4	0000	•	240.	566.	593.	848	366	183	69.	611.5
4	0000	•		40	3406.7	•8336	.2797	.1378	1.692	1344.49
5	0000		7795.	3170.	4078	7%	537	526	-	216.7
5	0000	•	6110.	2330.	3054	748	391	400	֚֚֡֡֡֜֝֜֜֝֜֜֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֡֓֓֓֓֓֓֡֓֓֓֓֡	705.2
S	0000	•	4069.	1181.	1706.	777	215	777		120.5
S	0000	•	231.	005.	370.	800	056	300	11.	430,0
3	0000	•	0385.	8707.	8937	820	897	347		156.6
S	0000	•	8723.	454.	576.	835	753	297	11.	201.00
S	0000	•	134.	193.	225.	848	616	247		327.1
5	0000	•	682.	954.	916.	852	490	197	111	964.5
000	10000.	0.0	437.		58.	848	383	153	113	645.6
an a	0000	•	20	894.	768.	.8276	.2956	.1156	2.115	1377.20
• 65	100001		9344	12171.1	2945	199	.671	4	.75	531.9
٠ ټ	0000	•	7064.	.0860	506.	818	474	438	75	879.8
9	0000	•	5051.	863.	0203.	833	.300	393	.75	305.R
9	0000	•	2996.	.999	8844.	847	.122	346	.75	763.2
	000	0.0	091.	500.	649	.8597	•	66	.75	274.9
φ.	0000	•	9324.	355.	297.	866	805	253	.75	826.1
•)))	•	639	198.	065.	865	099	207	2.750	2408.49

Σ×	ALT	DELT	SHP	-	T+FRES	ETAP	G G	C1	- 7	VFT
	000	•	128.	111.	915.	852	529	164	.75	036.
~		•	840	148	913.	827	418	125		712.8
.65	10000	0.0	3767.2	2301.8	2051.3	.7768	.3255	•0919	2.750	36.3
		i		•				۲	-	u
Σ. ×	_	_	SHP	_	+ + + + + + + + + +	ے م	٢	_	, 7	
4.	500	;	4286.	2104.	2983	650	.448	567	• 66	274.7
4	500	•	3477.	1731.	2521.	668	.366	549	99.	688.3
040	5000		2243.	1064.	1721.	663	.241	518	99.	613.0
040	5000	•	598	118.	616.	733	910	474	•66	162.8
. 4	5000	•	9134.	9089	9460	764	.926	426	•66	776.1
- 4	5000	•	657.	916	177.	794	776	371	•66	396.1
. 4	5000	•	310.	661.	829.	810	639	312	•66	054.2
. 4	500	•	052.	439.	526.	826	51		1.661	1737.97
· 4	5000	•	937.	320.	344.	842	399	202	• 66	453.2
40	15000.	0.0	2996.0	3271.7	3254.3	. 8385	03	153	• 66	207.0
	500	•	5146	1323.	2144.	717	.535	530	.07	461.
	500	•	4208	0877.	1588.	734	.440	509	.07	138.4
, K	5000	•	2654.	047.	598	762	.282	470	.07	688.4
'n	500	•	016.	9036	9433.	78	116	~	2.076	S
Š	500	•	9457	.696	238.	808	.958	373	• 07	832.6
	500		957.	859.	020.	827	806	321	.07	448.4
	5000	•	560.	755.	830.	842	665	269	.07	1.960
	5000	•	259.	667.	671.	851	533	218	•07	772.9
	500	•	110.	649	600	852	416	171	.07	485.9
.50	15000.	0	3158.4	2752.3	2664.5	.8364	.3202	129	•07	236.2
Σ×		_	I	-	+FRE	T A				WFT
	20000	0.0	12697.0	9628.2	10358.1	.7137	1.5197	.5327	2.036	3769.11

E ×	ALT	DELT	SHP	-	T+FRES	ETAP	a	CT	7	VF.T
	0000		2063	212	4	707	677	7	Č	0
		•	• • • •		• 000	1771	75.0	010	000	26000
		•	·/677	9 (1.	• 775	742	.340	4 90	• 03	251.0
S	0000	•	790.	044.	455.	773	.171	445	.03	860.5
	0000	•	506.	198.	493.	962	018	398	.03	519.8
S	0000	•	173.	231.	421.	817	.858	344	60.	178.0
ŝ	0000	•	958.	285.	396.	834	713	292	03	871.1
• 50	20000	0.0	-	36.	~		75	40	.03	586.3
5	0000	•	773.	413.	397.	851	451	188	.03	327.7
5	0000	•	2888.1	587	534.	Ø	.3457	.1431	2.036	1102.19
Σ×		T.	I		+FRE	TA	a a	-	7	FT
4	0000	ထံ	7512.	4058.	5112.	639	.569	582	.72	300.9
• 40	100001	18.0	16488.7	646.	-	.6595	7	65	7	37.5
4	0000	&	4813.	2820.	3584.	689	.327	531	.72	438.2
4	0000	æ	2937.	1788.	2375.	726	.159	488	.72	910.9
4	0000	æ	1211.	07070	1144.	761	.004	443	.72	427.2
	0000	œ	501.	9569.	875.	802	.851	396	.72	978.2
4	0000	œ	953.	116.	316.	813	712	336	.72	574.2
	0000	ထိ	486.	713.	825.	824	581	278	.72	197.9
	0000	ထိ	153.	.604	443.	836	461	224	.72	856.7
4	0000	œ	966	5	236.	848		1762	1.724	556
.50	0000	8	8536.	3181.	4159.	708	.661	5	.15	524.9
S	0000	æ	7283.	2634.	3468.	728	.548	523	.15	102.6
Ś	0000	œ	5300.	1633.	2260.	757	.371	482	.15	527.7
	100001	18.0	13445.1	10573.2	11033.6	.7833	1.2049	38	.15	002.3
W)	0000	œ	1602.	9390.	9702.	806	.039	389	.15	494.8
Ŝ	0000	80	9870.	165.	354.	823	.884	338	2,155	3043.01

Σ×	ALT	DELT	SHP		T+FRES	ETAP	G G	CT	- 7	VFT
.50	10000.	18.0	8255.0	6953.3	7044.1	.8390	. 7398	.2881	2.155	2625.63
5	0000	œ	759.	774.	780.	850	605	239	.15	244.4
	000	æ	372.	.009	544.	852	481	06	.15	894.0
S)	0000	æ	190.	556.	461.	45	375	147	•15	589.9
	0000	α	0428	2446	3327	48	6		α	7 660
· C	0000	, a	8510.	1504	2162	400	, v v	117	• C	366.5
•	0000	0	6331	368	810.	200	463	420	2 6	720.4
.65	10000	18.0	14339.8	926	9537.2	.8367	1.2851		80	~
9	0000	8	2361.	119.	244.	850	.107	336	.80	624.0
9	0000	8	0543.	012.	021.	861	446.	290	.80	155.8
•	0000	8	8838.	916.	833.	866	792	245	.80	722.1
9	0000	8	230.	817.	668.	862	648	199	.80	319.6
	0000	œ	812.	803.	595.	847	520	157	.80	965.5
9	0000	80	•909	914.	670.	819	412	120		9
	_	ij	I	-	+FRE	1		-	7	FT
	000	ө	1538.	5221.	6278.	716	.652	539	•19	393.5
	000	8	9327.	4216.	5044.	745	.482	504	.19	733.4
	000	8	7044.	2981.	3587.	772	307	460	•10	078.1
	000	8	4868.	1665.	2092.	795	.140	413	.19	465.8
• 50	5000	18.0		10251.7	523	815	.977	63	.19	887.7
	000	œ	0420	8869.	9015.	833	827	314	.19	379.8
	000	8	981.	484.	526.	844	689	265	.19	912.0
Ś	000	8	292.	129.	093.	852	559	217	.19	480.1
	000	.	792.	873.	769.	852	444	172	.19	7.960
5	000	80	530.	740.	04.	.8367	.3476	.1326	2.193	1766.87
3	F - 4	ũ	2	٠		,			•	
_ I	- (ָ . ה .	L	_ ;	-+FKEV	T A P	3	_	7	- X
• 20	15000.	œ	~	11273.1	2148	2		.5488	2.116	4703.55

Σ×	ALT	DELT	SHP	-	T+FRES	ETAP	a. O	CT	7	WFT
5	5000	œ	4864.	0923	1699.	718	. 565	بر 1	-	6
5	5000	ထိ	703	377.	024	740	7 7 7) R) C	4 -	0000
S	5000	æ	1981	9437	0012	770	. 261) U	→	06190
ĸ	5000	α	0470			- 6	•	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	770	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
ı u		•	•	. 000		2 Y C	501.	T. →	11.	120.7
5 1	2000	U	876	453.	651.	814	666.	361	11.	714.1
0	2000	œ	511.	387.	513.	837	791	311	.11	350.2
47 1	2000	ထ	173.	338.	385.	845	650	259	.11	6.600
• 50	15000.	18.0	σ	02.	4287.1	.8525	519	209	.11	8.669
n	2 00 0	œ	57.	357	291.	85	.4062	.1634	2.116	1424.93
	AL	E	I	-	+FRE	¥_	م	-		1
• 50	000	18.0	3108.	ю •	10319.1	66	.632	549	.07	961.1
R) I	00	ထိ	ω	9287.4	980.	713	54	34	.07	725.1
S	0000	ထိ	1790.	.696	579.	730	.468	516	.07	473.3
\mathbf{v}	0000	œ	0686.	387.	875.	753	.330	482	.07	139.5
	0000	သ	285.	554.	910.	781	.156	434	.07	747.6
	0000	œ	. 266	697.	945.	803	.995	385	.07	406.4
S	0000	ထိ	731.	777.	928.	823	838	332	.07	080.6
5	0000	œ	574.	873.	950.	839	969	280	.07	785.3
	0000	œ	440	973.	.066	851	557	228	.07	511.7
Ų	၁၀၀၀	o	511.	119.	087	S C	.4372	.1795	2.077	1268.62
	-	i		·						
	4	Ľ	Ι		+FRE	<u> </u>	ں	_		u
9	2000	•	6779.	665.	420.	793	.701	665	69	813.0
9	2000	•	5352.	9934.	0522.	807	.556	465	69	368.2
Ü	5000	•	3459.	897.	9291.	824	.364	417	69.	833.3
.65	15000.	0.0	11775.9	7924.6	8168.7	.8396	1.1938	71	9	376.2
9	5000	•	0068.	888	005.	853	.020	.3228	2.699	2937.11

F H	2542.81 2175.17 1836.56 1540.48 1289.14	5066.69 3963.08 3963.08 3625.08 2615.88 1884.65 1338.99	5184.05 4572.54 4032.07 3023.13 3073.19 2652.52 2265.70 1915.80 1612.48
7	2.699 2.699 2.699 2.699 2.699		
CT	.2763 .2296 .1830 .1408	. 4748 . 3862 . 3862 . 29407 . 2505 . 1628 . 0919	.4618 .4151 .3724 .3271 .2824 .2824 .1953 .1953
۵	.8633 .7137 .5732 .4519	1.8239 1.6284 1.4374 1.2520 1.0731 .9083 .4808	1.8836 1.6675 1.2638 1.1024 .9322 .7719 .6248
ETAP	.8637 .8681 .8618 .8408	. 810 . 824 . 624 . 624 . 855 . 855	.8143 .8270 .8376 .8465 .8510 .8495 .8495 .8197
T+FRES	5909.5 4829.6 3778.5 2832.7 2030.5	10837.4 9682.6 8534.5 7407.6 6303.0 5256.0 4228.1 3256.6 2399.4 1685.9	10517.2 9286.6 8189.3 7061.1 5980.6 4943.1 3949.9 3021.2 2210.7
-	5895.0 4898.0 3905.2 3003.8 2225.7	10131.5 9198.0 8240.5 7269.9 6292.5 5345.6 4395.1 3474.3 1960.7	9852.8 8857.7 7945.7 6980.2 6026.1 5086.7 4167.9 3289.9 2513.2
ЗНР	8515.8 7039.7 5654.1 4457.3 3468.1	17990.6 16062.3 14178.9 12349.5 10584.8 8959.5 7407.8 5966.7 4742.7	18580.3 16448.1 14567.9 12663.3 10874.5 9195.1 7614.5 6163.5 4909.4 F TABLE
DELT	00000	000000000	A N O O O O O O O O O O O O O O O O O O
ALT	15000. 15000. 15000. 15000.	15000. 15000. 15000. 15000. 15000. 15000.	15000. 15000. 15000. 15000. 15000. 15000. 15000.
Σ×		**************************************	m m m m m m m m m m m m m m m m m m m

T H	87.7 03.4 11.5	990.42 620.32 266.25 943.07	4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	45.2 90.6 07.2	18 86 18 86 18 86	86.57 371.39 149.33 739.95
7	.647 4 .647 3 .647 3	2.647 26 2.647 26 2.647 26 2.647 19	. 647 . 647 . 647 . 1	.054 .054 .054 .054 .054	8 8 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9	33.558 40 33.258 40 33.258 35 33.258 35 33.258 35 258 27
CT.	504 481 442	.3941 .3470 .2982 .2504	157 117	4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	2269 22669 1818 1401	.4413 .3945 .3520 .3054
a	693 596 439	1.2552 1.0660 .9190 .7646	488 488 488 488	835 706 503	11 12 12 12 12 12 13 13 13 13 13 13 13 13 13 13 13 13 13	1.9125 1.7541 1.5442 1.3601 1.1697
ETAP	789 798 813	. 8 310 8 4 78 8 5 8 8 8 5 6 8	851 821 821	806 815 830 830		.8099 .8195 .8322 .8431
T+FRES	813. 300. 442.	7413.4 6442.6 5458.7 4513.6	721. 9 75 .	430. 775. 741.	5836.8 4911.0 4007.5 3136.2 2343.9 1662.0	9269.1 8484.0 7449.7 6529.3 5556.3
-	122. 709. 002.	7121.9 6271.7 5389.3 4525.2	837. 122.	753. 235. 385.	2754 4100 3285 1874 1874 1874 8	8593.8 7975.0 7129.6 6362.1 5516.9
SHР	146. 340. 030.	10487.3 9073.3 7678.4 6387.8	163.	331. 257. 558.	9523.7 8093.6 6742.8 5462.9 4329.3	15979.0 14655.1 12901.5 11363.7 9772.5 8317.2
DELT	• • •		• • •			000000
ALT	0000	20000. 20000. 20000.	0000	00000	200000 200000 200000 200000	20000.
Σ×	999	0 0 0 0 4 0 0 0 0 0 0	9 9	トトトト	27. 27. 27. 27. 27. 27. 27. 27. 27. 27.	

Σ×	ALT	DELT	SHP	-	T+FRES	ETAP	G D	C1	7	FUZ
8. 08. 08.	20000. 20000. 20000.	000		3901.5 3121.7 2384.2	3766.4 2930.1 2159.7	.8489 .8353 .8037	.8285 .6737 .5347	.2159 .1727 .1319	3.258 3.258 3.258	2030.31 1718.71 1440.69
C) co	0000	ш О	7 A B L) 9	-245.5	. 3	.4208	0000000	3.258	1208.13
9	ALT 5000	• بـ	SHP 1749	T 703•	FRES 320.	786	CP •671	C1 506	J.	WFT 425.7
 	25000.	000	00	ψ œ		79	00	487	2.594	3211.32
9 4	5000		9258	341.	6666	621	.317	417		629.9
9	5000	•	9000	869	981.	852	974	320		003.2
9	5000	• •	664.	121. 375.	125. 348.	867	874 663	222	• 00° 0	461.0
	5000 5000	• •	85	65 98	8 8 8 8		50 4	.1308	. 0 0	225.3 017.8
	5000	ø	2773.	417.	035.	603	.817	487	66.	653.8
• 75	25000.	000	12107.0	7093.6	7622.9	.8222	1.7222	.4865	2.993	3412.04
1	5000	•	9728.	8.82	141.	836	.383	386	66.	717.2
- 1	5000	• •	207	103. 105.	340. 506.	857	.025	345 293	700	940.2
~	5000	•	045.	761.	727.	860	.860	247	66.	775.0
	5000	• •	934.	055. 379.	963.	830	702 557	2002	66.	505.6 261.7
7	5000	•	050	767.	608.	801	434	116	66.	050.7

Σ×	ALT	DELT	SHP	-	T+FRES	ETAP	d O	CT	-5	K F T
∞	5000	•	7 7 7 7	205	020	α α	0	0	•	6
œ	5000	•	610	0 0			000	101	, , ,	7.701
0		•		101	0	† †	\$ () t	なり	. 10	216.6
O	2000	•	1422.	391.	771.	825	.624	420	.19	159.7
∞	2000	•	.6000	681.	911.	837	.423	373	.19	770.0
œ	2000		709.	968	108.	846	.238	328	919	428.9
08.	00		420.	288.	296.	852	055	282	19	101.5
Œ	2000	•	212.	594.	525.	853	.883	236	910	801.0
œ	5000	•	082.	908.	783.	844	722	161	.19	529.2
ω	5000	0	053.	255.	088	821	576	148	19	282.9
ထ	5000	0.0	3167.5	1673.6	1482.9	.1797	.4506	.1101	3.192	1069.93
.85	5000	•	3951	204	2	809	984	473	39	918.8
œ	2000	•	100.	829.	348	817	.863	645	939	624.0
∞	5000	•	1698.	180.	527.	828	.664	406	939	210.3
œ	2000	•	0304.	514.	713.	839	.465	362	39	824.6
φ,	2000		945	828.	901.	846	.272	317	39	468.6
ر ت	2000	•	627.	129.	101.	848	085	271	39	136.2
0 0	5000	•	401.	450.	345.	845	.910	226	30	833.2
ω (2000		238.	778.	618.	831	745	182	939	553.9
	25000	0	84.	148.	950.	05	565	141	939	303.4
00	5000	•	284.	•	389.	92	.4672	.1058	3,392	1088.70
E ×	AL	_	SH	-	FRE	T	۵.	١.		u
9	0000	•	635.	437.	978.	784	.640	506	. 53	839.6
9	0000	•	185.	201.	682.	792	.564	488	53	667.7
	0000	•	700.	935.	354.	800	.481	467	53.3	493.2
.65	30000	0.0	8044.4	5567.5	6.6069	.8125	1.3699	38	.53	284.0
	၁ ၀ ၀	•	966	947.	181.	830	.191	3	2.539	2005.32

Σ×	ALT	DELT	SHP	-	T+FRES	ETAP	d O	C1	7	WFT
	000	•	038.	349.	98	845	028	342	53	755.6
•65	30000	0.0	46.	695.	765.	859	.859	291	53	05.8
9	0000	•	142.	064.	074.	898	705	241	50.00	282.6
	0000	•	301.	436.	400	866	562	191	5	077.R
9	0000	•	2546.6	1837.7	~	.8472	.4337	.1447	2.539	893.74
	0000	•	491.	207.	761.	801	.786	488	. 93	035.9
.75	30000	0.0	9666	967.	447.	808	702	469	.93	47.1
	0000	•	435.	691.	.760	817	•606	448	.93	648.9
~	0000	•	527.	217.	517.	828	.452	410	.93	379.5
	0000	•	399.	602.	785.	842	.260	362	. 93	080.1
	0000	•	358.	007.	.960	853	.082	315	.93	811.3
	0000	•	336.	394.	408.	861	908	267	.93	555.9
7	0000	•	389.	792.	747.	861	747	219	.93	324.2
	0000	•	507.	200.	114.	849	597	173	.93	112.1
1	0000	•	2720.1	1650.4	1534.7	.8219	.4632	.1299	2.930	923.17
• 80	0000	•	04 40	120.	84.	806	•		.12	145.9
	30000	0.0	.	872.	359.	812	779	462	.12	46.6
	0000		812.	571.	971.	820	.671	438	.12	728.0
	0000	•	730.	026.	301.	831	.486	395	.12	417.5
	0000	•	622.	447.	609	843	.298	350	.12	122.5
	0000	•	531.	853.	917.	852	.112	303	.12	842.2
	0000	•	• 565	255.	242.	856	.935	256	.12	582.7
	0000	•	522.	.999	595.	851	770	209	.12	346.4
	0000	•	628.	100.	986.	836	618	165	.12	132.3
œ	0000	•	813.	1558.3		.8002	6	.1227	3.125	938.86
.85	30000	0.0	11487.7	6054.7	6635.2	.8691	1.9563	.4767	3,321	3264.72
										•

Σ×	ALT	DELT	SHP	-	T+FRES	ETAP	CP	CT	7	VFT
00	0000	•	. 4000	700	0		ò	ì		
a		•	, ,	• • • • • • • • • • • • • • • • • • •	677	0 7 0	• 800	400	• 32	053.3
Ü		•	01/0	424.	844.	822	• 733	459	.32	803.3
X	0000	•	952.	868.	116.	834	.524	383	.32	458.3
œ	0000	•	856.	324.	460.	845	337	340	32	165.6
∞	0000	•	721.	724.	758.	850	.144	293	32	875.6
ထ	0000	•	673.	140.	.960	849	996.	247	32	613.0
. 85	30000	0.0	63.	•	59	42	794	201	.32	69.1
6 0	0000	•	742.	. 400	865.	822	637	157	.32	150.2
∞	0000	•	2919.0	1494.5	330.	• 7860	.4971	.1177	3.321	956.17
	⋖		SH	-	FR	-	a O	-		u
9	6809	•	424.	086.	5534.	782	.591	503	.47	721.7
9	6809	•	. 460	906	309.	290	.520	486	47	9-260
9	6086	•	747.	715.	072.	798	.445	467	747	4.640
9	6809	•	359.	492.	803.	807	362	445	. 47	827.0
9	6809	•	825.	177.	426.	819	248	413	47	670.0
Ŷ	6809	•	021.	676.	840.	836	920	364	47	450.1
9	6809	•	260.	177.	275.	852	.913	314	47	763.6
• 65	0	0.0	95.	644.	684.	864	652	262	47	74.1
9	6089	•	2791.1	2120.1	2118.5	.8680	.5981	.2100	2.472	405.67
• 75	36089.	0.0		74.	381.	800	.735	486	80	380.2
~	6809	•	740.	739.	151.	807	.658	469	85	239.2
~	6086	•	359.	544.	902.	814	.577	450	85	0.260
~	6086	•	901.	304.	605.	822	.478	426	8.5	0420
	608		•	3905.6	4123.7	34	322	386	80	738.0
~	6089	•	327.	425.	556.	847	.141	339	80	516.8
	6809	•	495.	928.	984.	.8589	.9634	.2901	2.852	1305.19

0 2982.2 1943.2 1902. 0 8478.5 4849.5 5329. 0 8678.7 4669.8 5092. 0 7182.3 4213.4 4508. 0 6322.0 3764.4 3963. 0 5492.5 3314.0 3427. 0 3068.0 1847.1 1787. 0 2393.8 1398.0 1308. 0 8692.4 4629.0 5063. 0 8661.1 4425.3 4795. 0 7456.7 4134.6 4423. 0 5655.8 3211.7 3307. 0 3186.0 1787.7 1766.8	•	(•	((
0.0 8478.5 4849.5 5329. 0.0 7182.3 4213.4 4508. 0.0 7182.3 4213.4 4508. 0.0 6322.0 3764.4 3963. 0.0 5492.5 3314.0 3427. 0.0 3828.4 2330.8 2311. 0.0 3868.0 1847.1 1787. 0.0 2393.8 1398.0 1308. 0.0 8492.4 4629.0 5063. 0.0 8492.4 4629.0 5063. 0.0 861.1 4425.3 4795. 0.0 6513.3 3664.4 3847. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3486.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	2.0 .8637 2.7 .8591	.6390	.2408	2.852	1112.44 936.52
0.0 8098.7 4669.8 5092. 0.0 7698.8 4477.4 4840. 0.0 7182.3 4213.4 4508. 0.0 6322.0 3764.4 3963. 0.0 5492.5 3314.0 3427. 0.0 3828.4 2330.8 2311. 0.0 3068.0 1847.1 1787. 0.0 2393.8 1398.0 1308. 0.0 8492.4 4629.0 5063. 0.0 8492.4 4629.0 5063. 0.0 8661.1 4425.3 4795. 0.0 86513.3 3664.4 3847. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3952.1 2250.0 2210. 0.0 2483.5 1338.1 1229.	.4 .8044 1	816	480	•	469.9
0.0 7698.8 4477.4 4840. 0.0 5492.3 4213.4 4508. 0.0 5492.5 3314.0 3427. 0.0 5492.5 3314.0 3427. 0.0 3828.4 2330.8 2311. 0.0 3828.4 2330.8 2311. 0.0 3868.0 1847.1 1787. 0.0 2393.8 1398.0 1308. 0.0 8492.4 4629.0 5063. 0.0 8492.4 4629.0 5063. 0.0 7456.7 4134.6 4423. 0.0 5655.8 3211.7 3307. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.6 .8109 1	735	462	.04	320.9
0.0 7182.3 4213.4 4508. 0.0 6322.0 3764.4 3963. 0.0 5492.5 3314.0 3427. 0.0 4624.3 2817.2 2853. 0.0 3828.4 2330.8 2311. 0.0 3068.0 1847.1 1787. 0.0 2393.8 1398.0 1308. 0.0 8864.2 4812.9 5312. 0.0 8651.1 4425.3 4795. 0.0 7456.7 4134.6 4423. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.6 .8179 1	649	443	.04	171.9
0.0 6322.0 3764.4 3963. 0.0 5492.5 3314.0 3427. 0.0 3828.4 2330.8 2311. 0.0 3068.0 1847.1 1787. 0.0 2393.8 1398.0 1308. 0.0 8884.2 4812.9 5312. 0.0 8651.1 4425.3 4795. 0.0 7456.7 4134.6 4423. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.8 .8250 1	539	417	• 04	2.000
0.0 5492.5 3314.0 3427. 0.0 4624.3 2817.2 2853. 0.0 3068.0 1847.1 1787. 0.0 2393.8 1398.0 1308. 0.0 8492.4 4629.0 5063. 0.0 8492.4 4629.0 5063. 0.0 8061.1 4425.3 4795. 0.0 6513.3 3664.4 3847. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3486.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.0 .8374 1	354	372	•04	766.4
0.0 4624.3 2817.2 2853. 0.0 3828.4 2330.8 2311. 0.0 3068.0 1847.1 1787. 0.0 2393.8 1398.0 1308. 0.0 8492.4 4629.0 5063. 0.0 8492.4 4629.0 5063. 0.0 8492.4 4629.0 5063. 0.0 8651.1 4425.3 4795. 0.0 7456.7 4134.6 4423. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.9 .8486 I	17	328	•04	48.3
0.0 3828.4 2330.8 2311.00.0 3068.0 1847.1 1787.00.0 2393.8 1398.0 1308.0 0.0 8492.4 4629.0 5063.00.0 8492.4 4629.0 5063.00.0 7456.7 4134.6 4423.00.0 5655.8 3211.7 3307.00.0 3952.1 2250.0 2210.00.0 3186.0 1787.7 1706.00.0 2483.5 1338.1 1229.00.0 2483.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 12488.0 1	.9 .8568	066	279	• 04	328.2
0.0 3068.0 1847.1 1787. 0.0 2393.8 1398.0 1308. 0.0 8884.2 4812.9 5312. 0.0 8492.4 4629.0 5063. 0.0 8061.1 4425.3 4795. 0.0 7456.7 4134.6 4423. 0.0 5655.8 3211.7 3307. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.9 .8563	820	230	• 04	132.6
0.0 2393.8 1398.0 1308. 0.0 8884.2 4812.9 5312. 0.0 8492.4 4629.0 5063. 0.0 7456.7 4134.6 4423. 0.0 6513.3 3664.4 3847. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.5 .8467	657	183	•04	952.4
0.0 8884.2 4812.9 5312. 0.0 8492.4 4629.0 5063. 0.0 8061.1 4425.3 4795. 0.0 7456.7 4134.6 4423. 0.0 5655.8 3211.7 3307. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.8 .8214	.5129	.1385		790.43
0.0 8884.2 4812.9 5312. 0.0 8492.4 4629.0 5063. 0.0 8061.1 4425.3 4795. 0.0 7456.7 4134.6 4423. 0.0 5655.8 3211.7 3307. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.					
0.0 8492.4 4629.0 5063. 0.0 8061.1 4425.3 4795. 0.0 7456.7 4134.6 4423. 0.0 6513.3 3664.4 3847. 0.0 5655.8 3211.7 3307. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.3 .8095 1	903	4	.23	567.0
0.0 8061.1 4425.3 4795. 0.0 7456.7 4134.6 4423. 0.0 6513.3 3664.4 3847. 0.0 5655.8 3211.7 3307. 0.0 4767.0 2720.0 2736. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.2 .8145 1	819	458	.23	411.2
0.0 7456.7 4134.6 4423. 0.0 6513.3 3664.4 3847. 0.0 5655.8 3211.7 3307. 0.0 4767.0 2720.0 2736. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.1 .8203 1	727	438	.23	252.0
0.0 6513.3 3664.4 3847. 0.0 5655.8 3211.7 3307. 0.0 4767.0 2720.0 2736. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.8 .8286 1	597	409	.23	57.1
0.0 5655.8 3211.7 3307. 0.0 4767.0 2720.0 2736. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.5 .8407 1	395	363	.23	804.0
0.0 4767.0 2720.0 2736. 0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229.	.1 .8486 1	211	318	.23	578.1
0.0 3952.1 2250.0 2210. 0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229. FIT SHP T THEFES	.5 .8526 1	021	269	.23	353.6
0.0 3186.0 1787.7 1706. 0.0 2483.5 1338.1 1229. FIT SHP T THEEES	.0 .8507	846	222	.23	153.8
0.0 2483.5 1338.1 1229.	.8 .8385	682	177	.23	970.€
3341 T GHV 113	.4 .8051	.5322	.1326	3,232	805.45
FIT CHP T TABLE					
	S ETA			~	u
. 0.0 6892.0 3957.0 4339.6	.6 .8072 1	.7800	.4725	3.041	2022.70

Σ×	ALT	DELT	SHP	-	T+FRES	ETAP	CP	CT.	7	H H
σ	0000	•	571.	805.	141.	814	697	403	.04	899.3
03.	40000°	0.0	4	3632.2	3918.5	.8205	9	4337	3.041	1772.55
α	0000	•	717	373.	597.	829	.476	402	40.	613.4
α	0000		954	967	107.	842	,279	354	40.	408.8
∞	0000	8	2640	583.	654.	8 5 5	101.	308	40.	227.9
α	0000	•	553,	166.	179,	857	116.	258	\$0	0.690
	0000	60	8969	760.	732.	\$ 12 \$ 12 \$ 4	748	220	40	888.5
Œ	0000	•	2840	361.	303	837	590	162	, 0%	41.6
8	0000	•		984.	906	798	48	7	• 04	0.4
8 8	0000		225	956	324.	811	86	468	23	102.
Œ	0000	•	896°	770.	115.	816	.781	500	23	972,9
Ċ	0000		5750	590.	880.	823	.682	\$ \$2 \$	• 23	836.1
α	0000	•	606	293.	505.	832	,526	30	53	651.4
	400004	0.0	°	92.	3020.3	.8438	\sim	03453	• 23	41.4
ထ	0000	•	396.	507,	562.	852	135	299	.23	252.1
œ	0000	•	670.	. 560	092.	853	.948	250	.23	6,690
	0000	•	966	.169	652.	846	774	202	.23	602.6
œ	0000		369.	310.	235.	826	612	156	.23	56.2
∞	0000	•	908	946	853.	782	466	113	3,231	.7
Σ×			I	-	FRE	TA		 	~	F
œ	5000	0.0	243.		3312.	.8122	.720	4		559.
نك	5000		686	905.	150.	818	.637	440	40.	461.7
∞	5000		662.	738.	939.	825	.529	415	• 04	350.5
• 80	0	•	46	440.	577	83	3	70	•	87.6
œ	5000	•	548.	142.	223.	849	.164	325	•04	040.3
8	2000	•	959.	804	834.	857	.971	273	• 04	889.9

FIR	756.66 632.27 519.81 424.13	621.57 517.44 392.12 216.15	4 5 8 8 6 9 6 9 6 9 6 9 6 9 6 9 6 9 6 9 6 9	WFT 970.49 798.67 625.57 439.57 175.51
7	3.041 3.041 3.041 3.041	33.2311		20000000000000000000000000000000000000
CT	.2241 .1743 .1265 .0851	.4558 .4362 .4087 .3619	000	.5272 .5272 .5087 .4896 .4659
G P	.7964 .6288 .4756 .3474	1.8068 1.7169 1.5935 1.3906	000 000 000 000 000 000 000 000 000 00	CP 1.7583 1.6806 1.6001 1.5050 1.3450
ETAP	.8559 .6432 .8089 .7454	.8152 .8210 .8286 .8410	0 0 0 4490	ETAP • 7778 • 7853 • 8031 • 8188
T+FRES	1469.7 1113.6 779.4 495.7 -71.1	3299.2 3128.3 2894.0 2514.5	769. 412. 066. 741. -76.	T+FRES 6981.0 6699.8 6405.3 6051.8 5449.0
-	1477.7 1149.4 834.0 561.3	3005.1 2875.7 2694.4 2386.0 2084.8	451. 432. 1114. 607.	T 6415.8 6191.0 5958.2 5670.1 5166.7
SHP	2427.3 1916.4 1449.6 1058.8 TABLE 746.2	6 3 5 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6		SHP 9892.6 9455.4 9002.9 8467.4 7567.7
DELT	0.0 0.0 0.0 0.0 0.0 RANGE OF		0.0 0.0 0.0 0.0 0.0 RANGE DF RANGE DF	DELT 18.0 18.0 18.0 18.0
ALT	45000. 45000. 45000. 45000. CP DUT DF	20000	00000000000000000000000000000000000000	ALT 30000. 30000. 30000. 30000.
£	ER R C R			× E • • • • • • • • • • • • • • • • • • •

Σ.	ALT	DELT	SHP	-	T+FRES	ETAP	ď	CT	7	F
	0000	œ	596.	968	072.	.8506	466	32	9.00	655.4
.65	30000	18.0	673.	361	398.	862	830	276	5	422.0
	0000	œ	811.	758.	744.	868	677	226	.59	208.5
	0000	æ	3015.2	2166.6	2115.1	61	•5359	œ	2.594	14
	0		,		ć	Ç	ć	((Ċ	1
	0000	ů	• 7 / / 0	1 20	900	ナバー	* * * *	0 0 0	7.7	0.11
~	0000	œ	292.	968.	483.	805	.829	064	66.	9886
/	0000	8	9779.	719.	162.	608	.738	410	66.	796.3
~	00	18.0	065.	372.	726.	820	611	441	66.	563.5
~	0000	æ	.646	792.	022.	834	.413	393	66.	246.3
~	0000	œ	938.	241.	375.	845	.233	348	66.	9.426
7	0000	œ	893.	645.	695.	856	.047	568	66.	707.8
7	0000	œ	949.	078.	063.	860	.879	253	66.	470.1
~	0000	œ	038.	503.	440.	858	717	205	66.	247.0
•75		•	3210.3	1952.1	1853.0	.8415	• 5706	•1604	2.993	1046.70
8	0000	80	1266.	.660	.969	66	.002	501	-	293.6
8	0000	œ	767.	884.	405.	806	.913	483	.19	2.960
œ	0000	80	0205.	621.	064.	813	.813	195	.19	889.2
• 80	30000	18.0	9339.	211.	544.	823	9	28	.19	18.5
œ	0000	8	174.	628.	836.	835	.453	380	.19	288.7
$\boldsymbol{\omega}$	0000	8	136.	087.	193.	845	.268	335	.10	010.7
œ	0000	æ	078.	511.	532.	852	.080	288	.19	740.7
œ	0000	8	105.	955.	912.	854	.907	242	• 19	496.8
ω	0000	8	170.	390.	299.	845	741	196	.19	0.692
œ	0000	œ	3322.1	1855.2	1730.2	.8243	.5905	.1524	10	065.
.85	30000	18.0	11794.8	6034.4	6650.3	.8024	2.0964	.4958	3.393	3419.26

Σ	ALT	DELT	SHP	-	T+FRES	ETAP	CP	CT	7	L HA
00	000	æ	1268.	808	341.	808	• 002	477	• 39	210.5
. 85	30000	18.0	637.	535	978.	816	890	454	99	983.4
α	0000	8	9589.	050	360.	826	.704	415	.39	6666.9
∞	0000	œ	423.	497.	682.	837	.497	369	• 39	334.9
∞	0000	œ	333.	956.	036.	846	.303	325	.39	045.8
	000	œ	253.	384.	378.	848	.111	278	.39	7.077
8	0000	œ	260.	836.	765.	845	.934	233	3,0	522.8
ထ	000	80	305.	290.	172.	834	765	188	.39	291.1
	0000	œ	3438.2	1772.3	1621.7	.8084	.6111	.1456	3.393	1083.89
Į	_	H	SI		FRE	V L	ပ	-		FT
æ	000	18.0	6543.	635.	9371.	801	.059	497	.32	734.3
œ	000	œ	5517.	193.	795.	811	.932	471	.32	379.0
	000	8	3866.	431.	842.	823	.726	427	.32	890.7
• 80	00	8	287.	676.	921.	834	529	384	.32	436.6
8	000	æ	0725.	902.	008	845	.335	339	.32	005.6
œ	000	8	9221.	105.	098.	850	.148	293	.32	618.2
œ	000	8	824.	329.	235.	849	.974	249	38	266.1
ထ	000	8	505.	575.	413.	844	809	205	.32	939.2
œ	000	8	279.	839.	632.	826	657	163	.32	642.7
	00	æ	4195.0	2168.0	1933.3	9	.5223	.1248	3.323	1380.34
Σ	_	ш	SH	-	FRES	T A	o O	-		F
œ	500	œ	3744.	300.	7973.	800	.037	500	.25	973.3
	500	æ	3078.	013.	587.	807	.938	480	.25	718.7
• 80	25000.	18.0	12156.5	6590.3	7046.4	.8166	1.8017	.4516	3.25R	3414.05
œ	500	8	0729.	910.	206.	829	.590	405	.25	9.400
8	500	8	9459.	281.	449.	841	.402	361	.25	643.3

X.	ALT	DELT	SHP	-	T+FRES	ETAP	C P	CT	7	T H
	5000	α	147		448	7	707	216	2.5	302.1
	25000	0.8	6947.7	30303	40006	8521	1.0297	2693	3.258	1997,30
		ά	700	276	170	25.0	250	226	0	712.3
o c		•	• / / -	- 0 0 0		0 0	, ,) 0) 0) F	, ,	
Ó	2000	œ	• 57/	• B 7 0	4 a D •	a a a		207	67.	4004
œ	5000	œ	757.	019.	845.	809	556	138	. 25	218.1
	_	u u	I	-	FR	T		-	7	u
•	0000	00	1266.	.660	6696.	662	.002	501	.19	293.6
	0000	œ	0767.	884.	405	806	.913	483	.19	2.960
ထ	30000	18.0	205.	621.	4	.8131	13	-	3.193	2889.26
	0000	æ	9339.	211.	544.	823	.660	428	.19	618.5
8	0000	8	174.	628.	836.	835	.453	380	.19	288.7
	0000	æ	136.	087.	193.	845	.268	335	.19	010.7
8	0000	œ	078.	511.	532.	852	.080	288	.19	740.7
∞	0000	œ	105.	955.	912.	854	.907	242	•19	496.8
∞	0000	8	170.	390.	299.	845	741	196	•19	269.0
.80	000	•	3322.1	1855.2	1730.2	82	• 5805		•10	065.0
Σ×	_	L.	I		FR	T A		_	7	FT
φ.	6809	ထိ	675.	819.	5319.	662	46.	499	.11	576.1
$\boldsymbol{\omega}$	6086	œ	309.	653.	.250	805	.862	482	.11	457.4
$\boldsymbol{\alpha}$	6809	8	923.	470.	858.	811	.776	463	.11	278.4
	6889	ထ	487.	264.	589.	819	879	441	.11	121.2
∞	6809	œ	776.	906	145.	829	.519	404	.11	905.5
\mathbf{w}	6809	ထ	887.	444.	589.	841	.319	357	111	7.499
	6089	œ	.090	.966	063.	851	.134	310	.11	448.9
• 80	36089.	18.0	4242.5	2525.5	2531.2	.8563	.9510	.2617	3.112	1243.28
	680	•	489.	.690	027.	853	782	214	.11	057.6

CT J WFT	.1676 3.112 886.28	CT J WFT 4912 3-111 2108-73 4729 3-111 1984-70 4279 3-111 1859-98 4279 3-111 1715-40 3375 3-111 1323-46 3375 3-111 1323-46 3939 3-111 821-28 11447 3-111 821-28 5654 446 5573-26 6554 446 5573-26 446 2334-36 446 1965-21 2825 -446 1965-21
9	.6224	11111111111111111111111111111111111111
ETAP	.8380	
T+FRES	1542.6	T+FRES 4132.2 4140.0 3937.6 3673.4 3228.3 2797.8 2797.8 1904.6 1182.7 1088.7 25480.7 25480.7 25480.7 25480.7 1088.7 11624.5 1624.5 1624.5 1624.5 1624.5 1624.5
-	1617.6	1 3932.4 3786.1 3633.2 3425.8 2701.0 1915.0 11527.8 11527.8 11527.8 11527.8 11527.8 11527.8 11527.8 11527.8 11527.8 11527.8 11527.8 1158.9
SHP	2776.5	2 SHP 6 734.9 6 734.9 6 412.0 5 984.6 5 269.9 4 5 91.3 3 2 19.3 2 2 6 10.7 2 0 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2
DELT	18.0	1188 00000 1188 00000000000000000000000
ALT	36089.	4 4 4 6 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
Σ×	• 80	

FHX	892 5625.85 892 4977.08 892 4330.91 892 3769.10 892 3254.67 892 2350.90 892 1980.43	338 6441.76 338 5697.84 338 5032.64 338 4384.50 338 3291.40 338 2291.40 338 2005.53	WFT 439 5522.66 439 5023.52
7	HH0047000		04
CT	00000000000000000000000000000000000000	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	.733 .733
CP	1.1990 1.0479 .8962 .7582 .5134 .3137	1.3931 1.02238 1.00692 0.00660 0.00692 0.00662 0.00662 0.00662 0.00662 0.00662 0.00662 0.00662	CP 1.3752 1.2515
ETAP	. 5082 . 5082 . 5499 . 5905 . 6316 . 7090	.5790 .6184 .6545 .7863 .7863 .8096	ETAP •2337 •2486
T+FRES	21912.2 20555.6 18891.7 17065.3 15099.3 10762.1 8672.5	21061.2 19547.8 17946.0 16176.0 14314.4 12292.1 10307.1 8333.5 6554.1	T+FRES 22189.9 21345.3
-	20709.4 19560.2 18100.8 16444.0 14625.0 12668.8 10524.1 8514.0	19778.7 18530.1 17135.4 15560.4 13854.6 11969.4 10100.5 8217.6 6503.3	T 20667.1 20001.9
SHP	18159.4 15871.1 13573.5 11483.0 9547.5 7775.6 6120.8 4751.5	21128.2 18533.9 16192.9 13878.7 11736.3 7944.9 6278.0 4892.1	SHP 17925.7 16312.8
DELT			DELT 18.0 18.0
ALT	00000000	0000000000	ALT 5000. 5000.
E ×	00000000	000000000	XM • 10

T	439 2512.42 439 2124.88 439 1780.89 439 1492.22	877 5595.57 877 5060.43 877 4460.18 877 3405.62 877 2951.40 877 2529.65 877 2139.72 877 1505.16	316 5711.09 316 5121.91 316 4520.72 316 3445.34 316 2984.90 316 2557.32 316 1816.30
7	& m =1 +0	HOOHOMOWOO	370344044 4444444
CT	. 505 . 440 . 370 . 303	00000000000000000000000000000000000000	00044880000000000000000000000000000000
a a	. 5488 . 4348 . 3357	1.4013 1.2661 1.1034 .9572 .6795 .5564 .4417	1.294 1.2916 1.1292 9.769 . 6944 . 5688 . 5688
ETAP	.4041 .4440 .4834 .5202	.4163 .4475 .4872 .5259 .6108 .6508 .7236	.5621 .5955 .6340 .7095 .7428 .7444 .8180
T+FRES	14730.6 12770.8 10697.6 8755.5	20111.4 19386.4 18237.3 16960.3 15430.9 13817.1 11990.3 10021.0	18596.6 17483.1 16129.4 14717.4 13058.6 11396.5 7864.9 6225.1
-	14260.7 12413.6 10435.5 8559.2	18753.3 18215.2 17284.2 16186.2 14819.6 13344.1 11641.3 9777.2 7940.9	17386.9 16485.4 15330.9 14094.3 12587.2 11056.2 7726.8
SHP	7153.6 5667.6 4376.0 3335.1	18265.7 16503.6 14383.0 12477.7 10564.5 7252.6 5757.2 4449.6	18811.8 16835.3 14705.9 12734.2 10789.8 7414.2 5895.5
DELT	118 118 18 18 18 18		
ALT	5000. 5000. 5000.		
Σ×	100	000000000000000000000000000000000000000	

Table I. - Propeller Design Information

DESIGN

```
Mach Number
                                                0.80
     Altitude
                                                11,000 km (36,089 ft.)
                                                5.740 Mw (7698.8 hp)
     Propeller Net Thrust
                                               19.6 kN (4414 lbf)
                                               1.615 kN (363 lbf)
     Residual Thrust
     Total Net Thrust
                                            = 21.25 kN (4777.6 lbf)
          P/D^2
                                               35.03
                                                0.817
          \eta_{\mathbf{p}}
          J
                                                3.081
                                                1.692
                                                0.4485
          Wf
                                            = 984.98 \text{ kg/hr} (2171.96 \text{ lbm/hr})
          TSFC
                                                0.046 kg/hr/N (0.455 lbm/hr/lbf)
                                                0.434
          \eta_{ov}
          T/P
                                                3.705 N/kw (0.621 lbf/hp)
     Propeller Diameter
                                                4.519 m (14.825 ft.)
                                            =
     Number of Blades
SLS/MAX THRUST
          P
                                               14.0 Mw (18,780 hp);
                                               maximum available = 15.23 Mw (20,424
                                               hp)
          Net Thrust
                                              88.79 kN (19,961 lbf)
          ^{\mathsf{C}}_{\mathsf{p}}
                                                0.8565
                                                0.5540
          J
                                                0.0
                                                1.940 Mg/hr (4277.9 lbm/hr)
          T/P
                                                6.342 N/kw (1.063 lbf/hp)
          TSFC
                                                0.022 kg/hr/N (0.214 lbm/hr/lbf)
          P/D^2
                                                490.1 kw/m^2 (61.067 hp/ft^2)
```

Note: Maximum sea-level-static (SLS) thrust does not occur at maximum engine horsepower because of propeller blade stall.

Table II. Comparison of Selected Engine Parameters

	Engine Thru	Engine Thrust - Weight Ratio		TSFC	۰ م
Engine Type	Sea Level Static	Sea Level	9144 m	urulse Perionmance (M = 0.8, 9144 m altitude)	m altitude)
(1) Turboprop (Propfan)	3.41	3.71	1.14	.456	.438
(2) Pratt and Whitney JT9D-25 Turbofan	5.50	4.22	1.43	.631	•309
(3) Pratt and Whitney STF-477 Turbofan	6.74	4.89	1.88	.575	.339

All engines are sized to produce net thrust equal to that of the JT9D-25 at M = 0.8, 9144 m altitude. Note:

THRUST SFC VERSUS UNINSTALLED NET THRUST STANDARD DAY

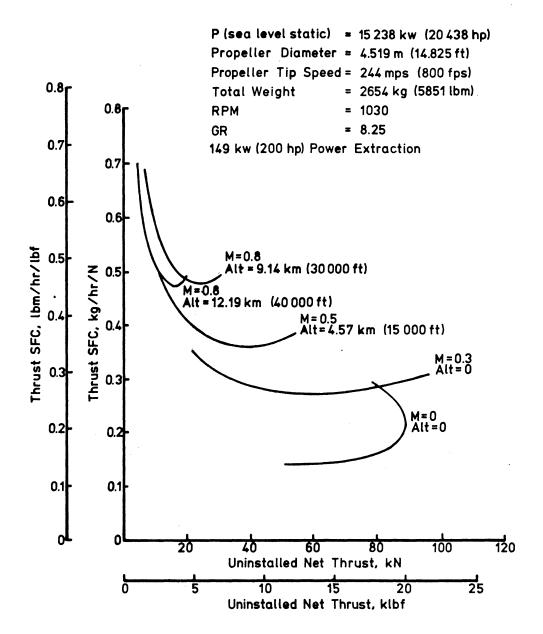


Figure 1. - Propfan performance.

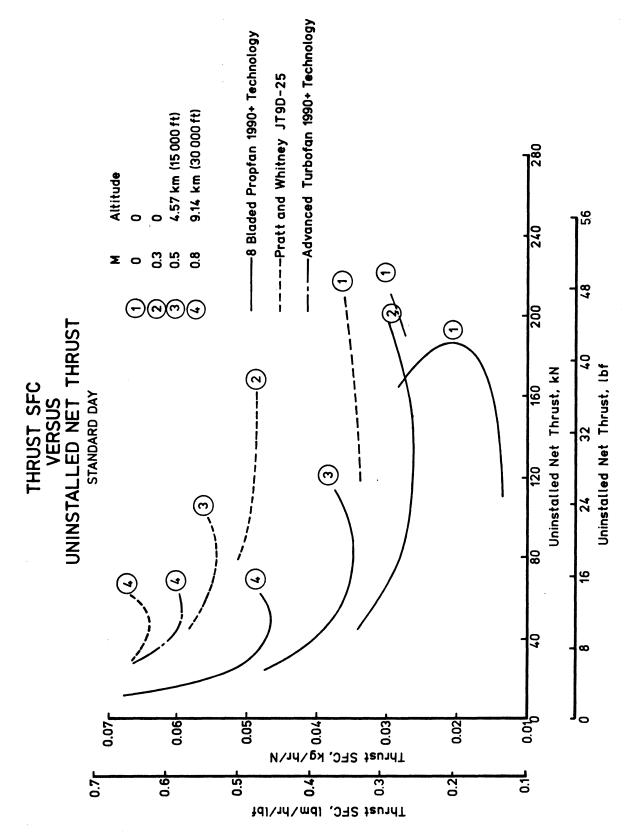
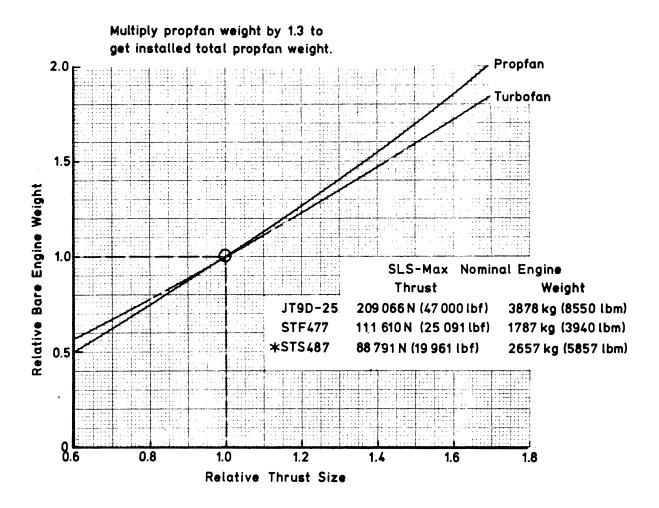


Figure 2. - Performance comparison



*Maximum thrust SLS not at maximum power because of propeller stall.

Figure 3. - Weight scaling for bare engine.

1.	Report No. TM-80075	2. Government Access	ion No.		3. Recip	eient's Catalog No.	
4.	Title and Subtitle Performance Estimation for Propeller Combined With a	or a Highly Loade an Advanced Techn	d Eight-E ology Tur	31ade rboshaft		rt Date 1979 rming Organization Code	
7.	Engine Author(s)		· · · · · · · · · · · · · · · · · · ·			rming Organization Report No.	
	Shelby J. Morris, Jr.			ļ			
9.	Performing Organization Name and Address					Unit No. -50-23-01	
	NASA Langley Research Ce Hampton, Virginia 23665			ŀ		ract or Grant No.	
12.	Sponsoring Agency Name and Address National Aeronautics and	Snace Administra	tion	<u> </u>		of Report and Period Covered	i
	Washington, DC 20546	Space Administra	CTOII		14. Spon	soring Agency Code	
15.	Supplementary Notes Pratt and Whitney repres indicated that this data	entatives have re i is neither class	viewed t ified no	he data r propri	in thi etary.	s report and have	
16.	Abstract						
	Performance estimation, propeller combined with useful for planned aircr Comparisons are made bet propulsion system and the 1990+ technology turbofa	an advanced turbo raft mission studi ween the performa ne performance of	shaft en es using nce of t both a c	gine are the tur he 1990+ urrent t	prese boprop techn echnol	nted. The data are propulsion system. ology turboprop	•
17.	Air Breathing Ga Propulsion Tu	ngines as Turbine urbofan		ion Statement		UNLIMITED	
	Propeller Turboshaft		Star Ca	tegory 0	7 - Ai <u>a</u> n	rcraft Propulsion d Power	
19.	Security Classif. (of this report) Unclassified	20. Security Classif. (of this Unclassified	page)	21. No. of 47	Pages	22. Price* \$4.50	